

# Global stability-based design optimization of truss structures using multiple objectives

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**Abstract.** This paper discusses the effect of global stability on the optimal size and shape of truss structures taking into account of a nonlinear critical load, truss weight and serviceability at the same time. The nonlinear critical load is computed by arc-length method. In order to increase the accuracy of the estimation of critical load (ignoring material nonlinearity), an eigenvalue analysis is implemented into the arc-length method. Furthermore, a pure pareto-ranking based multi-objective optimization model is employed for the design optimization of the truss structure with multiple objectives. The computational performance of the optimization model is increased by implementing an island model into its evolutionary search mechanism. The proposed design optimization approach is applied for both size and shape optimization of real world trusses including 101, 224 and 444 bars and successful in generating feasible designations in a large and complex design space. It is observed that the computational performance of pareto-ranking based island model is better than the pure pareto-ranking based model. Therefore, pareto-ranking based island model is recommended to optimize the design of truss structure possessing geometric nonlinearity.

**Keywords.** Nonlinear critical load; multi-objective optimization; island models; genetic algorithm; arc-length method.

## 1. Introduction

The stability strength of a truss structure is generally determined according to the magnitude of its linear buckling load computed by a local stability design method. Khot *et al* (1976) optimized the design of truss structures considering its linear buckling load. For this purpose, they utilized an optimality criteria approach in their design applications. In this regard, new optimization applications making use of the linear buckling load were also developed for structural design problems (Lin & Liu 1989). However, it was shown that a local stability analysis overestimates

the buckling load (Levy *et al* 2004). Therefore, the concept of global stability based on the computation of the nonlinear buckling load, also defined as *nonlinear critical load* or just *critical load*, was introduced into the stability-based design optimization (Levy *et al* 2004).

Plaut *et al* (1984) optimized the design of a small-scale truss taking into account of a critical load computed by use of a nonlinear buckling analysis. Khot & Kamat (1985) utilized a potential energy concept for their stability-based design optimization procedure. Kamat *et al* (1984) used an optimality criteria approach for optimal design of truss structures by imposing a uniform strain energy density to all truss members to obtain a maximum critical load.

Levy & Perng (1988) developed a two-phase optimization algorithm: a critical load was estimated for a specified external load; the truss structure was re-designed using an optimality criteria approach.

Levy (1994a and 1994b) showed that the values of member cross-sectional areas converged to a unique value at the end of optimization where geometric nonlinearity is taken into consideration.

Sedaghati & Tabarrok (2000) proposed an optimality criteria approach for truss structures which exhibit snap-through and snap-back behaviour. Their optimization approach was based on imposing a uniform strain energy density to all truss members.

Another challenging optimization approach was based on using the design sensitivity information, which is obtained from the nonlinear structural analysis, for optimal design of truss structures. Cardoso & Arora (1988) minimized the weight of truss structures utilizing its design sensitivity information. Choi & Santos (1987) and Santos & Choi (1988) obtained the design sensitivity information using the virtual work principle. Ohsaki (2001) proposed a similar approach based on the computation of sensitivity coefficients considering the critical loads of truss structures. Ohsaki & Ikeda (2006) attempted to compute the sensitivity coefficients considering some critical loads, which were indicated by bifurcated or branched points located in a load-displacement curve and defined as bifurcation point, degenerate critical point and hill-top branching point. Furthermore, they also showed that computation of sensitivity coefficients failed when iterative progression was terminated due to computational problem related to the singularity of determinants or when an inappropriate objective function was used to evaluate the sensitivity coefficients (Ohsaki 2005). In order to deal with this matter, certain joint deflections were constrained to move in specified directions at a predefined upper limit.

In this study, the design of truss structures is optimized considering the critical load which is computed by arc-length method based on iteratively adjusting the system rigidity matrix for tracing a load-displacement path. The arc-length method is also improved by including an eigenvalue analysis in its iterative algorithm in order to accurately estimate the critical load. Because of the negative effect of large joint deflections on the serviceability of truss structures, a new objective function related to joint displacements is included into optimization procedure. Although the weight of truss structure is an important factor for economical reasons, a higher critical load indicates larger stability strength for that truss structure. Therefore, a total of three objective functions are employed both to minimize the joint displacement and entire weight and to maximize the critical load of the truss structure. A pure pareto-ranking based multi-objective optimization model is employed in the optimization procedure. In order to increase the computational capacity of this optimization model, an island model, developed originally for the parallelization of evolutionary algorithms, is implemented into its search mechanism.

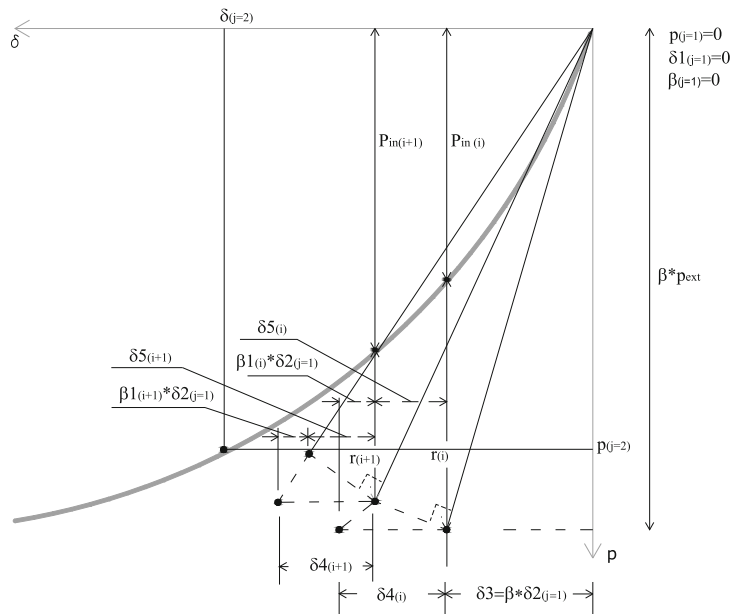
This paper is organized as follows. First, the fundamentals of arc-length method are presented. Next, the verification of arc-length method is demonstrated with two examples. Then, the pareto ranking-based multi-objective model and the island model are briefly described along with an introduction to the optimum design procedure. A search methodology is introduced for the

evaluation of the two multi-objective optimization models and execution of arc-length method. The discussion of results and the conclusions are also provided.

## 2. Arc-length method

The ultimate-load carrying capacity of a truss structure is determined by estimating the critical loads. But, the estimation of critical loads is difficult when a truss structure exhibits snap-through or snap-back behaviour. A discontinuity on the load-displacement path arises when the system rigidity matrix becomes indefinite; this prevents a further iterative progression in the estimation of critical loads. Therefore, the efficiency of any nonlinear solution method is measured by its capability of estimating all critical points. Arc-length method is proven to be efficient in the estimation of both the bifurcation and branching points (Kouhia & Mikkola 1999). Thus, arc-length method is improved by implementing new strategies in its computational procedure. (Summary of recent improvements for the arc-length method can be seen in references (Kouhia & Mikkola 1999; Memon & Su 2004; Ritto-Correa & Camotim 2008).

One such effective improvement is based on utilizing a constraint equation for the determination of incremental nodal displacements; hence a successful tracing is provided for the load-displacement path (Crisfield 1997). However, the constraint equation is represented by a quadratic form; when an inappropriate root is used for the constraint equation, the computational procedure fails. In order to overcome this difficulty, Krenk (1995) developed an alternative technique, called *orthogonal residual method*. The orthogonal residual method is based on adjusting a load increment for iterative procedures in a way of checking the orthogonality of residual force to current displacement increment (see figure 1). In this study, Krenk's arc-length method is



**Figure 1.** Iterative and incremental processes performing on load-deflection path.

utilized to estimate the critical load (see figure 2). The formulation of element stiffness matrix for a truss bar is presented in Magnusson (2006).

Some determinant-related criteria are of importance for the computation of critical loads by arc-length method. These are

- (i) Determinant of tangent stiffness matrix,  $\det[\mathbf{K}]$ .
- (ii) Minimum pivot of tangent stiffness matrix,  $\min\_piv[\mathbf{K}]$ .
- (iii) Minimum eigenvalue of tangent stiffness matrix,  $\min\_eig[\mathbf{K}]$ .

Although there are different possibilities of how the critical points are computed by the determinant-related criteria, the choice of  $\det[\mathbf{K}]$  may be inappropriate either due to the possibility of missing some points that represent the critical loads or due to the numerical problems associated with the computer program (Rezaiee-Rojand & Vejdari-Nogreiyani 2006). Therefore,

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Initialization of some parameters for incremental and iterative stage

p = [0]
δ1 = [0]
β = 1

Stage of increment

for j = 1: inc_max

1i. Compute [K] using δ1
1ii. δ2 = [K]-1 * pext
1iii. Normalize δ2j for first incremental stage and calculate β:
      β = (normalize(δ2j=1)) / (normalize(δ2))
1iv. Calculate δ3 for beginning of iterative procedure:
      δ3 = β * δ2

Iterative Stage

for i = 1: it_max

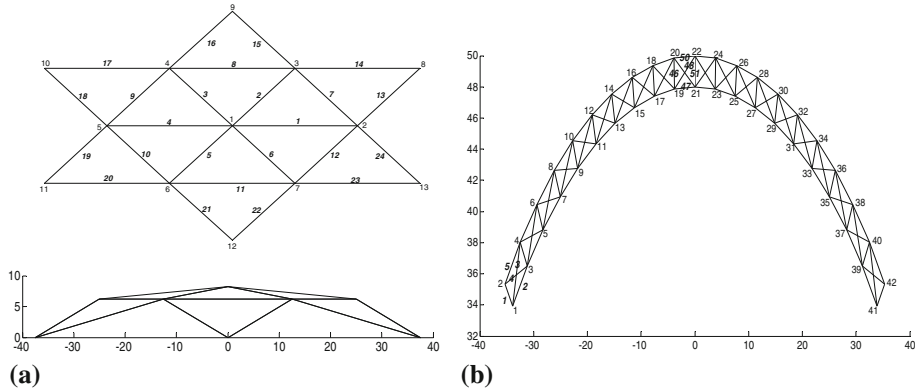
2i. Compute pint using δ1 + δ3
2ii. r = β * pext + pint - p
2iii. Update [K] using r
2iv. δ4 = [K]-1 * r
2v. β1 = -(δ4 * δ3 / δ3 * δ2)
2vi. δ5 = δ4 + β1 * δ2
2vii. δ3 = δ3 + δ5
2viii. β = β + β1

        if normalize pin ≤ ε * pext
            STOP
        else
            δ2 = δ2/2, δ3 = δ2, pext = pext / 2
        end
    end

1v. δ1 = δ1 + δ3
1vi. p = p + β * pext
end

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**Figure 2.** A pseudo code for proposed arc-length method.

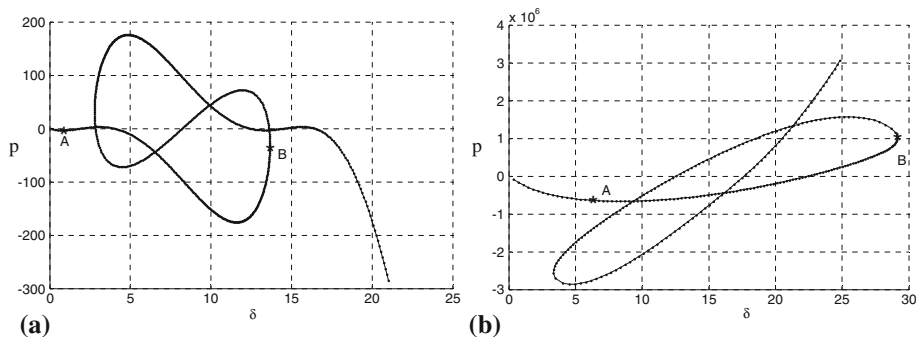


**Figure 3.** Mesh and geometry attributes of 24 (a) and 101 bar truss structures (b).

the best choice used as a determinant-related criterion seems to be  $\min_{\text{eig}}[K]$ . For this purpose, an interpolation process is utilized to determine zero eigenvalues. Eigenvalues of the tangent stiffness matrix are monitored until a sign change in an eigenvalue is noticed. Then, the displacement values are interpolated between the two values corresponding to these oppositely-signed eigenvalues in order to find a displacement corresponding to the zero eigenvalue. It must be noted that the point corresponding to a zero eigenvalue may not be located on the fundamental equilibrium path.

### 2.1 Demonstration of arc-length method by verification examples

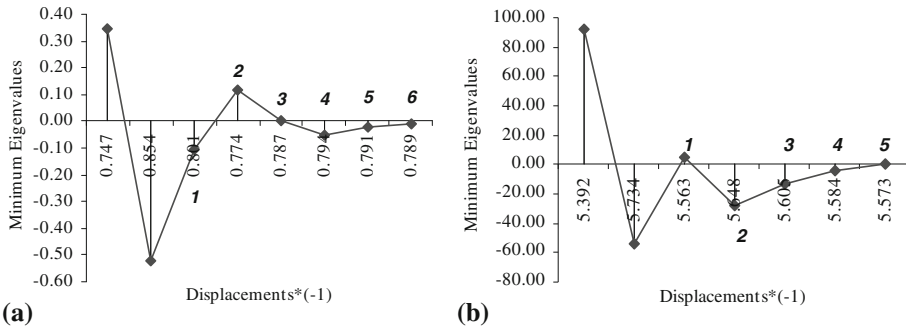
In order to demonstrate the arc-length procedure with eigenvalue analysis, two truss structures with 24 and 101 bars are analysed. The 24 bar truss structure, whose elasticity modulus and cross-sectional areas are taken as 10796 psi and  $1 \text{ in}^2$  respectively, is a star-shaped dome with a vertical load applied at its crown (figure 3a) (Wrigger *et al* 1988). The second verification example is a planar arch with 101 bars, which is also vertically loaded at its crown (Crisfield



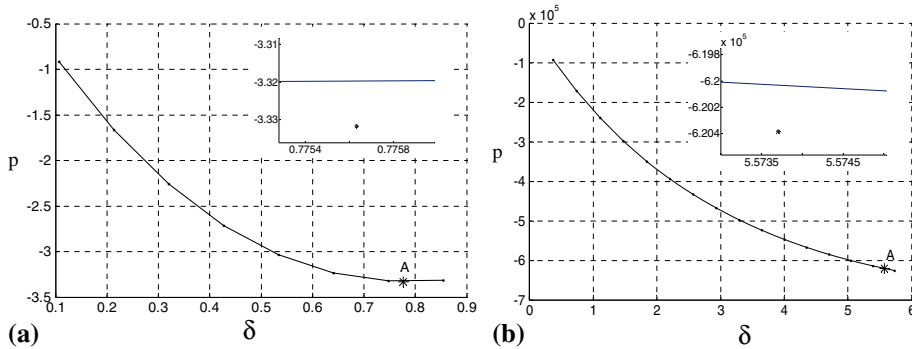
**Figure 4.** Post buckling plot of nonlinear responses for 24 (a) and 101 bar truss structures (b).

**Table 1.** Output from execution of proposed arc-length method for 24 and 101 bar truss structures.

	24 bar truss structure	101 bar truss structure
The outcome from a complete run of arc-length procedure		
Load corresponding to point B (kips)	30.989	1006478.352
Disp. corresponding to point B (in.)	13.657	29.139
Elapsed time for (in second)	10.30	20.12
Total number of increment	87	154
The outcome from computation of critical load and corresponding displacement		
Critical load corresponding to point A (kips)	3.332	620384.207
Critical disp. corresponding to point A (in.)	0.7756	5.5739
Elapsed time (in second)	1.98	2.24
Total number of increment	11	19



**Figure 5.** Variation on both minimum eigenvalues and displacements including corresponding iteration number for 24 (a) and 101 bar truss structures (b).



**Figure 6.** Locations of critical load and corresponding displacement for 24 (a) and 101 bar truss structures (b).

1997; Gien 2007). Its members have an elasticity modulus of  $5 \times 10^7$  psi and a cross-sectional area of  $1 \text{ in}^2$  (figure 3b).

The computational procedure of the proposed arc-length method is executed allowing a maximum error of 0.001. According to the results obtained, a load-displacement curve exhibiting snap-through and snap-back behaviour is simultaneously plotted for both truss structures (figures 4a and b). In this post-buckling graph, the values of load and displacement corresponding to point B are summarized together with the computation time in table 1. It is observed that the results obtained agree with those obtained with other approaches in Crisfield (1997); Gien (2007).

In the second step, two sequential points with oppositely-signed eigenvalues are determined. Then, an interpolation process between these two points is carried out until a zero eigenvalue is obtained. The interpolation process is illustrated for both examples in figures 5a and b. Figures 6a and b display the exact values of critical loads and related displacements (point A on the graphs) corresponding to zero eigenvalue. The corresponding computation times and total increment numbers are shown in table 1.

### 3. The solution of multiple objectives by island models

Multi-objective optimization models (MOMs) vary with respect to two key issues: sampling of feasible solutions from a large and complex search space and assessing them according to convergence degrees of their optimal designations. A MOM is performed using a set of solutions (called *pareto optimal set*) simultaneously. In this regard, at each run of evolutionary optimization algorithm, the objective is to obtain a solution, named *pareto solution*, satisfying conditions of constraints lying in the feasible region. In this regard, a *pareto front* is represented by the pareto solutions, each of which is not dominated by the other ones and represented by a curve, called *true pareto front*. Points located on the true pareto front curve are not improved further. The quality of *current pareto front* obtained in the end of a whole genetic search is assessed according to its closeness to the true pareto front.

Evolutionary algorithms have a flexible search mechanism. In particular, they have the ability to hybridize with a pareto-ranking based MOM (PbMOM) through both letting the pareto optimal set to be handled easily and allowing the combination of multiple objectives into a single objective function (Veldhuizen & Lamont 1998). The most popular evolutionary algorithm is the genetic algorithm. The genetic algorithm is governed by genetic operators which mimics selection and recombination within populations in nature. The genetic algorithm has been improved by new model implementations. One such model used in combination with the genetic algorithm is the island model that is developed for parallelization of genetic algorithms (Cantu-Paz 1999). It is managed by multiple populations called *islands* and categorized under coarse-grained models of parallel genetic algorithm. Coarse-grained models are managed by relatively few islands, each of which have a large number of individuals. The main strength of island models is their ability of handling multiple populations with different genetic parameters at the same time.

The islands are capable of exchanging their individuals through a transforming process called *migration* (Eby *et al* 1999). Migration operator is governed by a predetermined communication topology that prescribes which islands can interchange individuals. Migrating individuals (emigrants or immigrants) are determined depending on their fitness quality. The other essential parameters of migration operator are the number of emigrants or immigrants (migration rate) and the number of executions of the migration process (migration interval).

In order to increase the computational performance of island models, some attempts, based on adjusting parameter values of genetic operators, have been made Eby *et al* (1999) and proposed an island model with a fixed migration frequency. Malott *et al* (1996) used a coarse island model with slowly varying migration rates. One variation involves a *competition* which is based on reshuffling of islands with regard to their individuals' fitness values (Schlierkamp-Voosen and Mühlenbein 1996). The essential parameters of competition operator are the competition rate and interval.

#### 4. Design problem

In order to solve the optimization problem for the geometrically nonlinear truss structures, three objective functions are proposed. The maximization of limit load  $L(X)$ , the minimization of truss weight  $W(X)$  and joint deflection  $\delta(X)$ . Design variables  $X$  contain cross-sectional areas  $A$ , joint coordinates  $x$ ,  $y$  and  $z$  of truss structure. It is noted that additional design variables, for example radius of arch, distance between joints, etc. are used to define the shape of truss structures. The problem can be formulated mathematically as

$$\left. \begin{aligned} X &\in A_i, x_j, y_j, z_j && (i = 1, \dots, nda), (j = 1, \dots, nn) \\ \max L(X) &= \max (\text{critical load capacity corresponding to state of } \det[K] = 0) \\ \min \delta(X) &= \min (\delta_m) && (m = 1, \dots, nn) \\ \min W(X) &= \min \left( \sum_{m=1}^{nm} A_m * L_m * \rho \right) && (m = 1, \dots, ntm), \end{aligned} \right\} \quad (1)$$

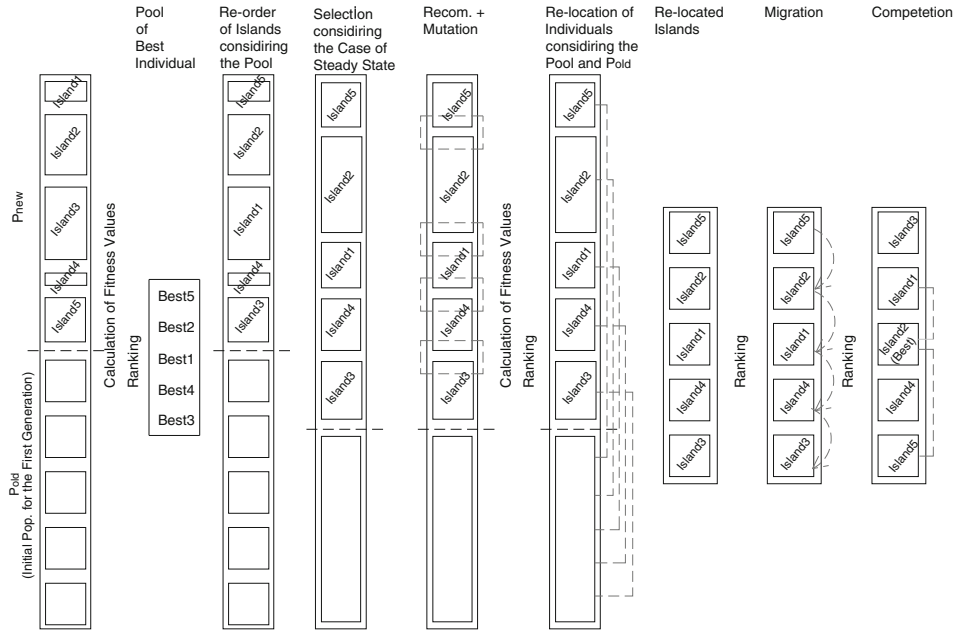
where  $\rho$ ,  $nda$ ,  $nn$  and  $ntm$  represent material density, number of different cross-sectional areas, nodes and truss members, respectively.

#### 5. Optimum design procedure

Islands models have the ability of investigating the different regions of complex search space, thereby using different parameter values for the same genetic operators. They need to be modified for multi-objective optimization procedure. In this study, all pareto optimal solutions computed through different objective functions are combined into a single pareto optimal set and then ranked. These ranked values are used to both arrange the order of islands and redistribute the individuals to different islands. The proposed algorithm, named pareto-ranking based multi-objective island model (PbMIM), is detailed in figure 7.

- (i) Initialize the islands by assigning randomly-generated numbers to its individuals which of chromosomes are coded using design variables of continuous type.
- (ii) Specify a combination set, which contains the cross-sectional areas and joint coordinates.
- (iii) Execute the proposed arc-length method and compute objective functions.
- (iv) Execute the ranking process for values of combined fitness functions. If required, apply the ranking share procedure based on re-scaling of fitness values according to their rank (Goldberg & Richardson 1987).
- (v) Select the individuals with higher computational performance according their fitness values and store in the 'pool of best individuals'.
- (vi) Re-order the islands in a descending order with respect to fitness values (see figure 7).





**Figure 7.** Depiction of the proposed island model for the design of multi-objective optimization procedure.

- (vii) Mate a randomly-chosen individuals located in different islands and apply crossover operator at pre-determined rates; choose randomly an individual and apply mutation operator at predetermined rate (see parameters values of genetic operators in table 2) (Polheim 1998).
- (viii) Select the individuals using a selection operator (see table 2) and re-create both the current island considering these individuals and the pool of best individuals.
- (ix) Migrate the individuals to the related islands using a ring-shaped migration topology (table 2) and a migration policy which is determined by ‘the best fitness value’ obtained from the pool of best individuals.
- (x) Assign the individuals, determined by the competition operator, to the related islands depending on the competition rate.
- (xi) Replace the initial island with the current island and repeat steps 2–10 until the predetermined number of generations is completed.

The computational procedure of PbMOM contains only the steps 1–4. Following the step 4, the individuals with higher computational performance are selected according to their fitness values and used to create the next population. These steps are repeated until a pre-determined number of generations is completed. Therefore, the search mechanism of PbMOM is considerably simpler than PbMIM’s.

## 6. Search methodology

According to the traditional search methodology, the computational performances of MOMs are assessed considering closeness of their current pareto fronts obtained to a true pareto front known beforehand (Talaslioglu 2010). This is accomplished by utilizing the quality measuring



metrics, which are computed using the optimal designations obtained. The accuracy of these quality measuring metrics must be confirmed through statistical tests. Hence, a statistical test for the evaluation of these quality-measuring metrics must be performed with a certain level of confidence. Therefore, a reasonable approach is to obtain a true pareto front by runs of PbMOM and PbMIM in bigger and repeated generation numbers. Optimal designations obtained are utilized in computation of the quality-measuring metrics. Details of the quality-measuring metrics and related statistical tests are presented in following sub-sections.

### 6.1 Quality-measuring metrics

Differentiation in MOMs architecture makes it difficult to lay down the different aspects of MOMs' computational performance. Therefore, quality-measuring metrics play an important role in the accurate prediction of MOMs' computational performance. In this study, two fundamental quality-measuring metrics, inverted generational distance and spread are employed.

6.1a *Inverted generational distance*: Inverted generational distance (Igd) estimates the far of non-dominated solutions included in current pareto front generated by the proposed MOM, from those included in true pareto front (see Eq (2)).

$$IGD = \frac{\sqrt{\sum_{i=1}^n d_i^2}}{n}, \quad (2)$$

where  $n$  is number of non-dominated solutions found by proposed MOM and  $d_i$  is Euclidian distance between each of these and nearest member of true pareto front. A lower value of Igd indicates a better approximation of current pareto front obtained to the true pareto front in terms of convergence.

6.1b *Spread*: This metric is used to measure an expanding spread exhibited by non-dominated solutions obtained and computed as,

$$S = \frac{d_f + d_l + \sum_{i=1}^{N-1} |d_i - \bar{d}|}{d_f + d_l + (N - 1) * \bar{d}}, \quad (3)$$

where  $d_i$  is Euclidian distance between consecutive non-dominated solutions,  $\bar{d}$  is mean of these distances,  $d_f$  and  $d_l$  are distances to extreme solutions of current pareto front. A lower  $S$  value implies points out a better distribution among non-dominated solutions. In other words, it is implied that non-dominated solutions are located in different positions.

### 6.2 Statistical tests

After computing means and standard deviations of quality-measuring metrics, a statistical analysis is performed in a certain level of confidence. If a probability value resulted from a statistical test satisfies a user defined significance level, then it is said that distribution of PbMOM and PbMIM approximation set is acceptable.

The statistical analysis is performed using MATLAB (The MathWorks, Inc., Natick, MA, 2008) software. Firstly, the outcomes related to quality-measuring metrics are checked through

**Table 3.** Optimal designations obtained by execution of PbMIM for arc structure with 101 bar using four design variable sets.

	Case I					Case II					Case III					Case IV				
Size var. (in <sup>2</sup> )	AREA1	160.9626	AREA1	30.8523	AREA6	30.5792	AREA6	1.0000	AREA6	30.5792	AREA6	1.0000	AREA6	30.5792	AREA6	1.0000	AREA6	30.5792	AREA6	1.0000
	AREA2	15.6072	AREA2	97.3767	AREA7	78.3764	AREA7	1.0000	AREA7	78.3764	AREA7	1.0000	AREA7	78.3764	AREA7	1.0000	AREA7	78.3764	AREA7	1.0000
	AREA3	18.5337	AREA3	150.0000	AREA8	57.6079	AREA8	1.0000	AREA8	57.6079	AREA8	1.0000	AREA8	57.6079	AREA8	1.0000	AREA8	57.6079	AREA8	1.0000
	AREA4	18.1993	AREA4	1.0000																
	AREA5	2.5747	AREA5	57.1093																
Shape variables (in)	R1	47.0527	R1 = R2 = R3 = R4 = R5	42.4947	R1	45.2512	R1 = R2 = R3 = R4 = R5	41.8930	R1	45.2512	R1 = R2 = R3 = R4 = R5	41.8930	R1	45.2512	R1 = R2 = R3 = R4 = R5	41.8930	R1	45.2512	R1 = R2 = R3 = R4 = R5	41.8930
	R2	45.6594	R6 = R7 = R8 = R9	40.0000	R2	43.1667	R6 = R7 = R8 = R9	40.0000	R2	43.1667	R6 = R7 = R8 = R9	40.0000	R2	43.1667	R6 = R7 = R8 = R9	40.0000	R2	43.1667	R6 = R7 = R8 = R9	40.0000
	R3	45.1627	R10 = R11	49.0274	R3	44.7358	R10 = R11	41.1860	R3	44.7358	R10 = R11	41.1860	R3	44.7358	R10 = R11	41.1860	R3	44.7358	R10 = R11	41.1860
	R4	44.0024			R4	45.4489			R4	45.4489			R4	45.4489			R4	45.4489		
	R5	44.2600			R5	43.2976			R5	43.2976			R5	43.2976			R5	43.2976		
	R6	44.5517			R6	44.2321			R6	44.2321			R6	44.2321			R6	44.2321		
	R7	44.3901			R7	42.6177			R7	42.6177			R7	42.6177			R7	42.6177		
	R8	46.0038			R8	45.6489			R8	45.6489			R8	45.6489			R8	45.6489		
	R9	46.0255			R9	46.2084			R9	46.2084			R9	46.2084			R9	46.2084		
	R10	48.8256			R10	45.4815			R10	45.4815			R10	45.4815			R10	45.4815		
	R11	49.0584			R11	47.8314			R11	47.8314			R11	47.8314			R11	47.8314		
	a1	2.3120	a1 = a2 = a3 = a4 = a5	2.6199	a1	1.0599	a1 = a2 = a3 = a4 = a5	2.0124	a1	1.0599	a1 = a2 = a3 = a4 = a5	2.0124	a1	1.0599	a1 = a2 = a3 = a4 = a5	2.0124	a1	1.0599	a1 = a2 = a3 = a4 = a5	2.0124
	a2	2.4052	a6 = a7 = a8 = a9	1.1669	a2	1.0449	a6 = a7 = a8 = a9	3.0000	a2	1.0449	a6 = a7 = a8 = a9	3.0000	a2	1.0449	a6 = a7 = a8 = a9	3.0000	a2	1.0449	a6 = a7 = a8 = a9	3.0000
	a3	1.5546	a10 = a11	0.5824	a3	2.4704	a10 = a11	1.5546	a3	2.4704	a10 = a11	1.5546	a3	2.4704	a10 = a11	1.5546	a3	2.4704	a10 = a11	1.5546
	a4	1.8712			a4	1.8656			a4	1.8656			a4	1.8656			a4	1.8656		
	a5	1.1247			a5	1.7117			a5	1.7117			a5	1.7117			a5	1.7117		
	a6	0.9859			a6	1.6628			a6	1.6628			a6	1.6628			a6	1.6628		
	a7	1.5207			a7	1.4512			a7	1.4512			a7	1.4512			a7	1.4512		
	a8	1.9310			a8	1.3849			a8	1.3849			a8	1.3849			a8	1.3849		
	a9	2.0794			a9	2.8820			a9	2.8820			a9	2.8820			a9	2.8820		
	a10	2.1887			a10	1.8825			a10	1.8825			a10	1.8825			a10	1.8825		
	a11	1.0528			a11	1.0148			a11	1.0148			a11	1.0148			a11	1.0148		

Table 3. (Continued).

	Case I	Case II	Case III	Case IV
$\beta_1$	2.8800	ANGLE1	11.9462	ANGLE1
$\beta_2$	6.2481	ANGLE2	15.8403	ANGLE2
$\beta_3$	10.6027			
$\beta_4$	15.4941			
$\beta_5$	20.1528			
$\beta_6$	23.7544			
$\beta_7$	29.4886			
$\beta_8$	34.6977			
$\beta_9$	37.4972			
$\beta_{10}$	42.0105			
			$\beta_1$	3.0236
			$\beta_2$	7.2061
			$\beta_3$	10.7937
			$\beta_4$	17.1034
			$\beta_5$	22.1458
			$\beta_6$	26.2338
			$\beta_7$	31.3397
			$\beta_8$	32.0000
			$\beta_9$	37.6140
			$\beta_{10}$	43.5137
			AREA1 = A1 = A6 = A11 = A16 = A21 = A26 = A31 = A36 = A41 = A46 = A51	
			AREA2 = A2 = A7 = A12 = A17 = A22 = A27 = A32 = A37 = A42 = A47	
			AREA3 = A3 = A8 = A13 = A18 = A23 = A28 = A33 = A38 = A43 = A48	
			AREA4 = A4 = A9 = A14 = A19 = A24 = A29 = A34 = A39 = A44 = A49	
			AREA5 = A5 = A10 = A15 = A20 = A25 = A30 = A35 = A40 = A45 = A50	
			AREA6 = A1 = A2 = A3 = A4 = .....A18 = A19 = A20 = A21	
			AREA7 = A22 = A23 = A24 = A25 = .....A38 = A39 = A40 = A41	
			AREA8 = A42 = A43 = A44 = A45 = .....A46 = A47 = A48 = A49	
			$\beta_1 = (\text{ANGLE1}/5)$ , $\beta_2 = (\text{ANGLE1}^*2/5)$ , $\beta_3 = (\text{ANGLE1}^*3/5)$ , $\beta_4 = (\text{ANGLE1}^*4/5)$ , $\beta_5 = \text{ANGLE1}$	
			$\beta_6 = \beta_4 + \text{ANGLE2}^*(1/4)$ , $\beta_7 = \beta_4 + \text{ANGLE2}^*(2/4)$ , $\beta_8 = \beta_4 + \text{ANGLE2}^*(3/4)$ , $\beta_9 = \beta_4 + \text{ANGLE2}$ , $\beta_{10} = (45 + \beta_9)/2$	

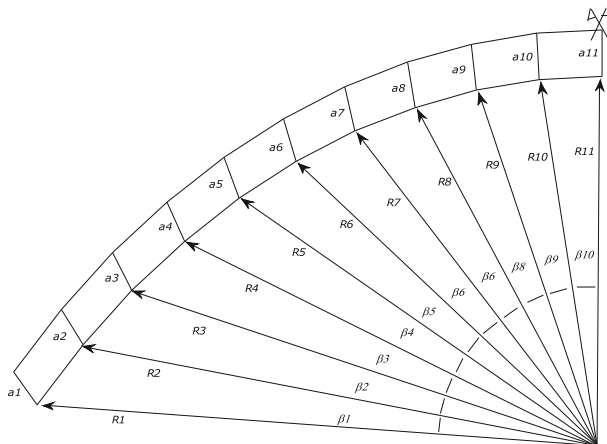
KolmogorovSmirnov test to inspect it whether to exhibit a normal distribution at 5% (0.05) significance level. If the variance turns out to be homogeneous, an Anova test is performed; otherwise, a Welch test is utilized (Ortiz & Walls 2003).

## 7. Discussion of results

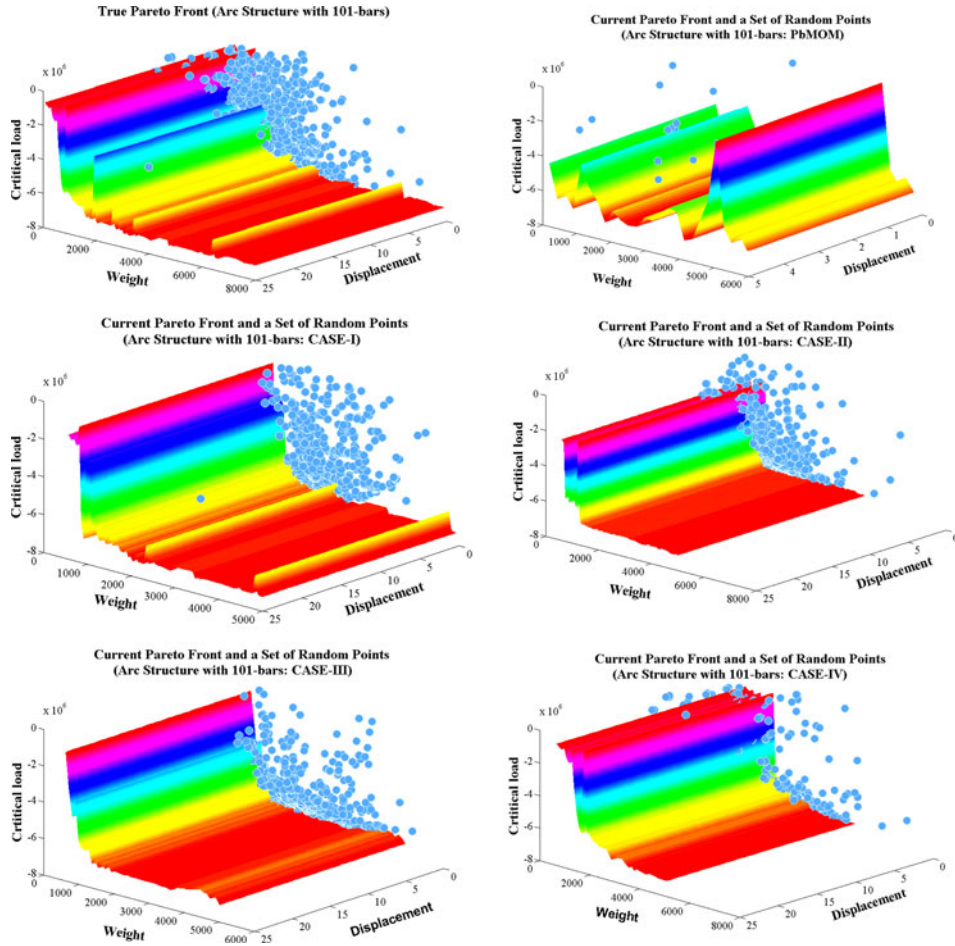
The application examples are chosen among the real-world truss applications, each of which has a large number of truss members and a large span. The examples include an arch structure with 101 bars, a pyramid structure with 224 bars and a dome structure with 444 bars. The designs of these truss structures are optimized considering the multiple objectives without imposing any constraint. The design optimization of these truss structures is carried out by use of both PbMIM and PbMOM to compare their relative computational performances. For this purpose, considering the search methodology mentioned above, a true pareto front is obtained for each design example along with their current pareto fronts and random point sets. Then, the closeness of their current pareto fronts to the true pareto front is measured taking the quality-measuring metrics into account. These quality-measuring metrics is checked by the statistical testing procedure mentioned above. The successful multi-objective model is used for further examination of its optimal designations.

In order to consider the shape effect on design optimization, the size and shape-related design variables are simultaneously included in the optimal design of truss structures. Several combination sets of these design variables are devised for the size and shape-related design variables. A maximum joint deflection of ‘max. span/300’, a serviceability requirement prescribed by AISC (American Institute of Steel Construction), is also considered for evaluation of the optimal designations. The modulus of elasticity and density of the truss material is taken as  $50 \times 10^6$  lb/in<sup>2</sup> and 0.1 lb/in<sup>3</sup>.

The parameter set assigned in the arc-length method, namely ‘number of increment’, ‘the number of iteration’, and ‘desired convergence degree’, is taken, respectively, to be (75, 75, 0.001) based on experience gained from several trial runs. Since the arc-length method is managed by an iterative-based computational procedure, the magnitude of load is adjusted at each



**Figure 8.** Geometrical parameters used to form arc shape.



**Figure 9.** True pareto front and current pareto fronts obtained by PbMIM (containing case I–IV) and PbMOM for design example of arc structure with 101 bars.

**Table 4.** Statistical test results computed by use of quality-measuring metric values (arc structure with 101 bar).

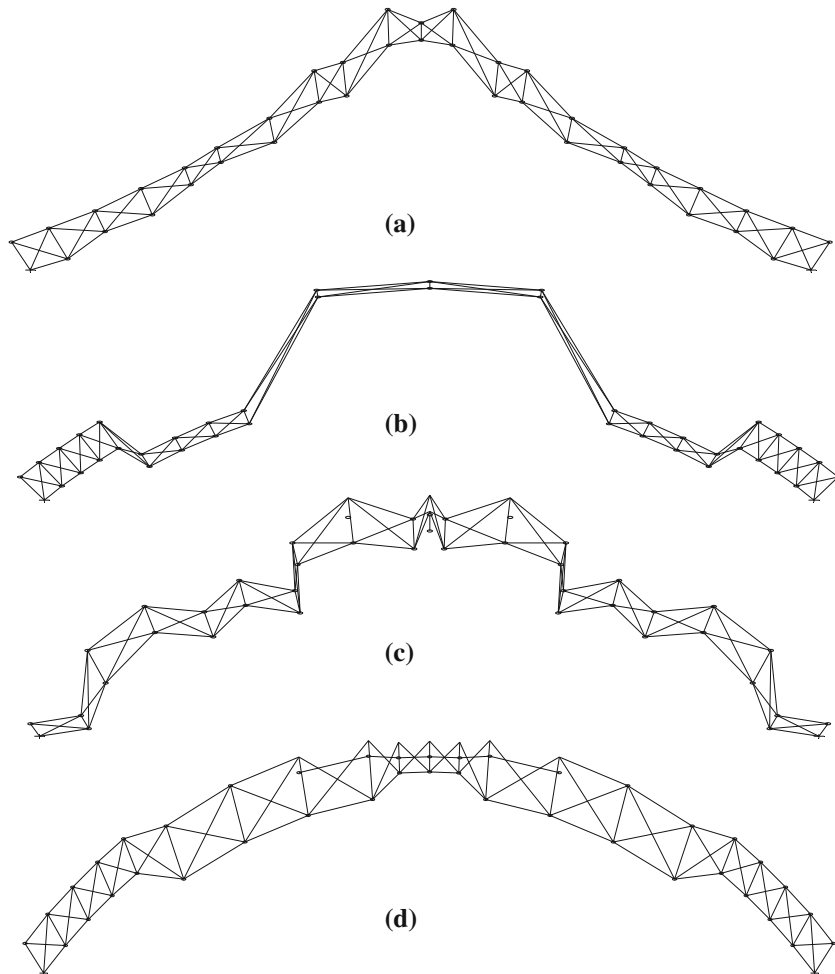
Multi-objective model	Case no.	Mean (spread)	Std (spread)	P (Welch test)	Mean (Igd)	Std (Igd)	P (Welch test)
PbMIM	1	0.840	0.067	0.930	0.145	0.086	0.518
	2	0.867	0.059		0.210	0.171	
	3	0.854	0.080		0.199	0.069	
	4	0.873	0.082		0.218	0.108	
PbMOM	1	0.875	0.055		0.259	0.093	

Mean: average of spread and Igd. Std: standard deviation of spread and Igd

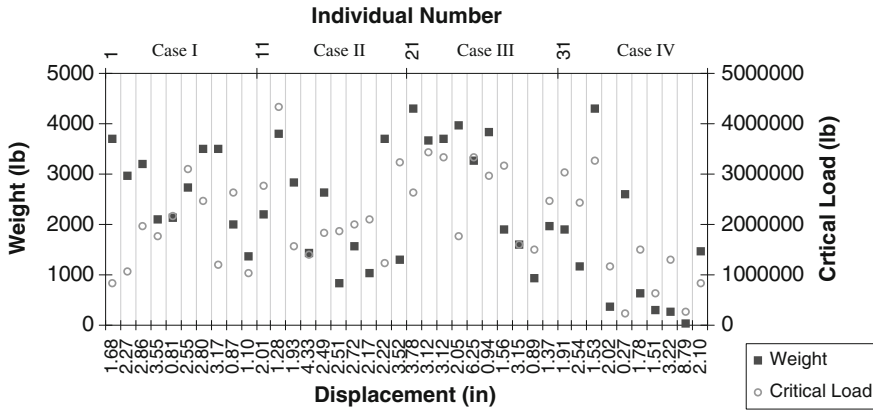
**Table 5.** Genetic output obtained by execution of PbMIM for arc structure with 101 bar (see table 3).

		Case I	Case II	Case III	Case IV
Min. weight (lb)	(Fitness fun. 1)	1017.8264	2690.9995	1974.8400	32.8973
Min disp (in)	(Fitness fun. 2)*	0.2180	0.9026	0.3190	5.6317
Max. crit. load (lb)	(Fitness fun. 3)	2044363.4099	1066399.8177	1597224.3750	696366.7552
No. of increments		19	20	25	23
No. of iteration for last increment		10	12	14	12
Mean fitness Values (lb)	Fitness fun 1	2307.2134	2025.6656	2631.5167	1123.2807
	Fitness fun 2	2.4115	2.8443	2.9010	2.7825
	Fitness fun 3	1883125.5520	1873942.1765	2242967.7266	1422018.8589
Standard deviation of fitness values	Fitness fun 1	839.2990	1389.0367	1282.8348	1599.1873
	Fitness fun 2	2.0594	1.7065	1.7740	2.2449
	Fitness fun 3	851568.5236	897190.8531	899561.2363	937744.2370

\*Computed at apex of arc structure

**Figure 10.** Optimal shapes of arc structure with 101 bar obtained by use of design variables represented by case I (a), case II (b), case III (c), case IV (d).





**Figure 11.** Variation of displacement and critical load at apex with weight of arc structure with 101 bar.

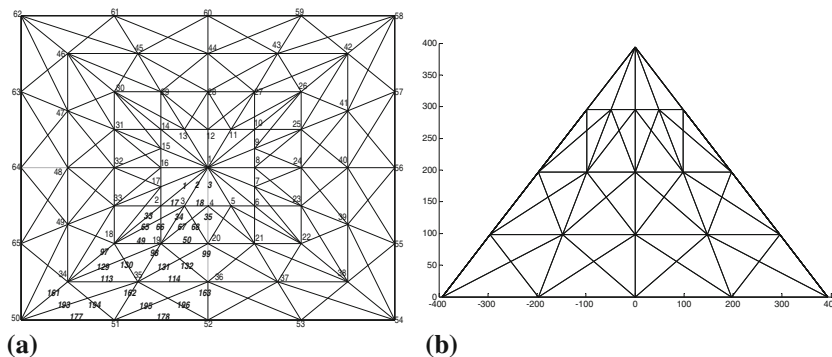
iteration according to the incremental procedure (see section 2). Therefore, the magnitudes of loads imposed to truss examples are not presented.

The computational optimization involving the optimization procedure and the structural analysis is coded within MATLAB software.

### 7.1 Application example 1: Arc structure with 101 bar

This arch structure was firstly used to test the computational performance of Crisfield’ arc-length method (Crisfield 1997). In this study, it is tackled to both verify the accuracy of proposed arc-length method and demonstrate the efficiency of PbMIM and PbMOM with various design variable sets. This truss has a vertical load at the apex (figure 3b).

The cross-sectional area of each member, denoted by  $A$  is used to represent ‘size-related design variable’. The ‘shape-related design variables’ are the ‘radius of arch segments’  $R$ , the ‘length of radial member of the arch’  $a$  and the ‘angle between two sequential arch segments’  $\beta$ . Four different sets of design variables are represented by Case I–IV (table 3). The shape-related design variables  $AREA1$ – $AREA8$  are also presented in table 3. The left symmetric part



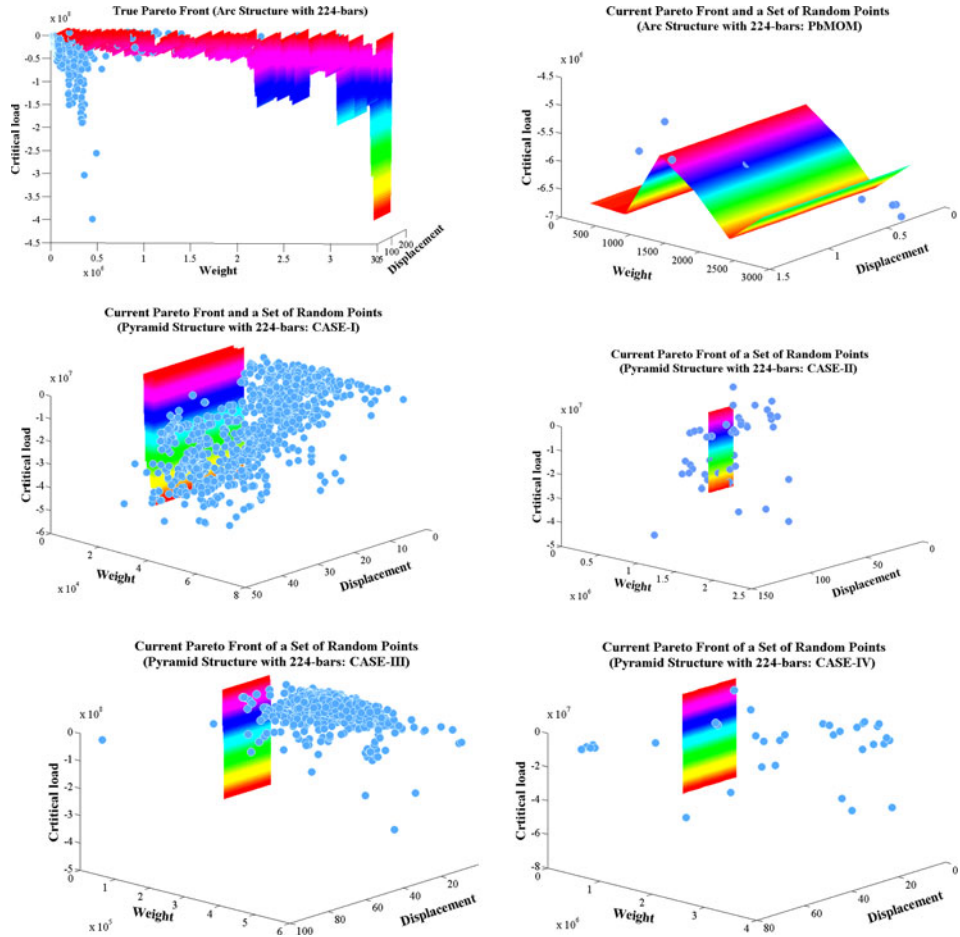
**Figure 12.** Mesh (a) and geometry attributes (b) of pyramid structure with 224 bar.

**Table 6.** Optimal designations obtained by execution of PbMIM for pyramid structure with 224 bar using four design variable sets.

	Case I			Case II			Case III			Case IV		
Size variables (in <sup>2</sup> )	A1-A16	24.2942	A1-A16	149.0759	A1-A32	51.5805	A1-A32	150.0000	A1-A32	150.0000		
	A17-A32	17.6759	A17-A32	29.6939	A33-A96	4.7873	A33-A96	115.3339	A33-A96	115.3339		
	A33-A48	7.1223	A33-A48	147.3168	A97-A160	1.3723	A97-A160	101.4316	A97-A160	101.4316		
	A49-A64	19.1728	A49-A64	96.9014	A161-A224	11.7474	A161-A224	150.0000	A161-A224	150.0000		
	A65-A96	3.0117	A65-A96	100.1490								
	A97-A112	21.8630	A97-A112	78.5355								
	A113-A128	10.3537	A113-A128	2.7681								
	A129-A160	9.8436	A129-A160	54.0079								
	A161-A176	8.2299	A161-A176	99.7886								
	A177-A192	8.1202	A177-A192	53.9105								
	A193-A224	22.2440	A193-A224	77.4478								
	X2 = Y2	87.1397	COOR4	107.5644	X2 = Y2	93.4627	COOR4	19.6900	COOR4	19.6900		
	X3	27.4220	COOR1	304.5414	X3	32.2971	COOR1	374.0200	COOR1	374.0200		
	Y3	79.8973	COOR5	131.5565	Y3	96.2233	COOR5	118.1100	COOR5	118.1100		
Y4	85.4722	COOR2	222.4209	Y4	96.1116	COOR2	255.9100	COOR2	255.9100			
COOR1	274.0398	COOR6	294.4615	Z2-3-4	271.5141	COOR6	216.5400	COOR6	216.5400			
X18 = Y18	169.6411	COOR3	137.3684	X18 = Y18	196.8500	COOR3	19.6900	COOR3	19.6900			
Y19	31.2537	COOR7	364.4316	Y19	64.7037	COOR7	314.9600	COOR7	314.9600			
Y20	174.9412	Z1	471.1836	Y20	122.2735	Z1	457.0870	Z1	457.0870			
COOR2	136.7108			Y20	160.5842							
X34 = Y34	169.0793			Z18-19-20	123.6876							
X35	217.6667			X34 = Y34	193.7063							
Y35	65.7197			X35	118.2568							
Y36	263.1244			Y35	281.4179							
	293.9940			Y36	238.7530							

**Table 6.** (Continued).

Case I	Case II	Case III	Case IV
COOR3	87.6198	Z34-35-36	63.6610
X50 = Y50	303,4166	X50 = Y50	216.5400
X51	155,4822	X51	150.2354
X55	357,4782	X55	356.4779
Z1	361,9844	Z1	383.8143
	X1 = X4 = X12 = X20 = X28 = X36 = X44 = X52 = X60 = 0.000		
	Y1 = Y8 = Y24 = Y40 = = Y56 = Y16 = Y32 = Y48 = Y64 = 0.000		
COOR1 = Z2 = Z3 = Z4			COOR5 = X18 = Y18 = X19*2 = Y19 = Y20
COOR2 = Z18 = Z19 = Z20			COOR6 = X34 = Y34 = X35*2 = Y35 = Y36
COOR3 = Z34 = Z35 = Z36			COOR7 = X50 = Y50 = X51*2 = Y51 = Y52
COOR4 = X2 = Y2 = X3*2 = Y3 = Y4			



**Figure 13.** True pareto fronts and random point sets for design example of pyramid structure with 224 bars.

**Table 7.** Statistical test results computed by use of quality measuring metric values (arc structure with 224 bar).

Multi-objective model	Case no.	Mean (spread)	Std (spread)	P (Welch test)	Mean (Igd)	Std (Igd)	P (Welch test)
PbMIM	1	0.821	0.080	0.338	0.089	0.057	0.145
	2	0.843	0.083		0.127	0.108	
	3	0.827	0.037		0.115	0.075	
	4	0.880	0.070		0.155	0.071	
PbMOM	1	0.893	0.058		0.198	0.055	

Mean: average of spread and Igd. Std: standard deviation of spread and Igd

of this arch structure is presented to display both size and shape-related design variables of arch structure (figure 8). The (half) arch includes 10 segments, each of which contains eleven  $R$  (denoted by  $R1-R11$ ), ten  $\beta$  values (denoted by  $\beta1-\beta10$ ), eleven  $a$  values (denoted by  $a1-a11$ ), and two independent angle values  $ANGLE1$  and  $ANGLE2$  (see Appendix A for the details on computation of nodal coordinates).

In order to evaluate the computational performance of PbMIM and PbMOM, their true pareto front and current pareto fronts obtained are presented in figure 9. Also, a statistical output obtained by computing the values of quality-measuring metrics is tabulated in table 4. Considering table 4, there is no considerable difference among spread and Igd values of PbMIM and PbMOM due to satisfying condition as ( $p > 0.050$ ). Furthermore, the Spread and Igd values of PbMIM are lower than PbMOM. Therefore, the quality degree of optimal designations corresponding to PbMIM is higher than PbMOM.

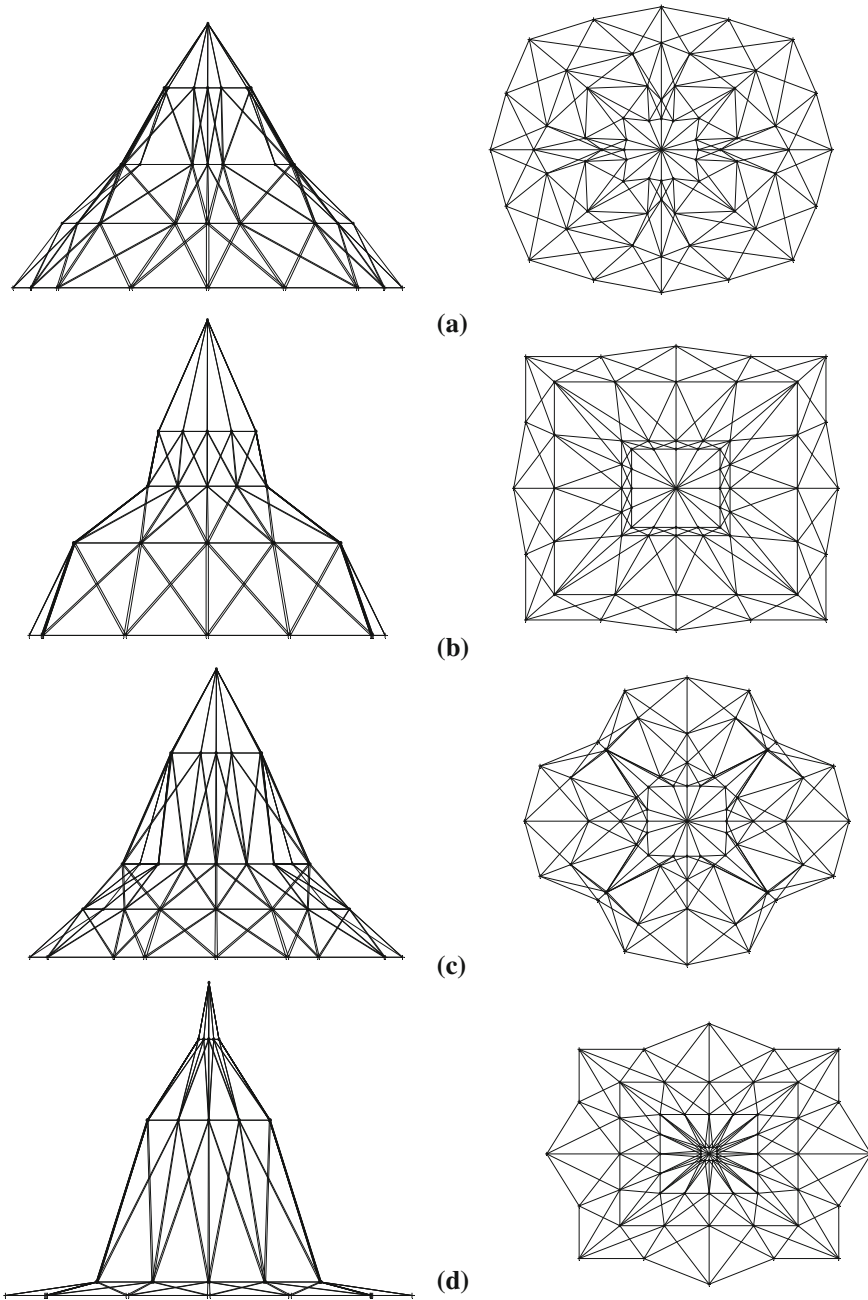
Particularly, the success order of four cases obtained by use of PbMIM is Case I, III, II and IV. In this regard, some designations obtained by PbMIM are picked from the related cases to carry out a further observation about them. Case I achieves to obtain a better optimal designation with a displacement of 0.2180 in. at node 1 along the x-direction and a critical load of 2044363.4099 lb (table 5) thereby satisfying the serviceability requirement prescribed by AISC ( $34.901 \times 2/300 = 0.2327$ ). It is apparent that the poorest fitness values set (32.8973 lb, 5.6317 in. and 696366.7552 lb) corresponds to Case IV (see table 5). Thus, final shape of arch structure corresponding to Case IV is obtained to be similar to its initial shape (figure 10d). Considering figure 11 and table 5, it can be said that the convergence degree of optimal designations corresponding to Case IV is lower than Case I.

Examining the arch member cross-sectional areas, it is recognized that there is a direct relation between the cross-sectional areas of diagonal members and the quality degree of optimal solution. In Case I, cross-sectional area values of diagonal members (160.9626, 18.5337, and 18.993 for members collected in the group called 'AREA1-AREA3') are higher than cross-sectional area values of the remaining members (15.6072 and 2.5747 for members collected in the group called 'AREA4-AREA5'). Furthermore, member cross-sectional areas have a tendency to increase as they approach the arch support points. Same results are observed for Case II (table 3).

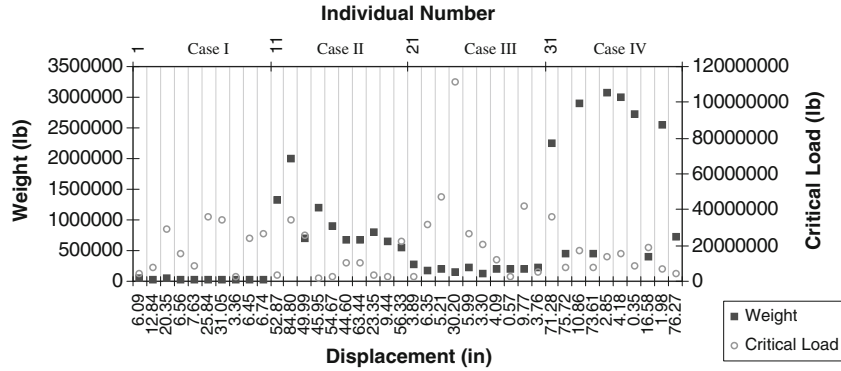
**Table 8.** Genetic output obtained by execution of PbMIM for pyramid structure with 224 bar (see table 6).

		Case I	Case II	Case III	Case IV
Min. weight (lb)	(Fitness fun. 1)	42284.5605	282165.9660	34624.2193	442358.8447
Min. disp. (in)	(Fitness fun. 2)*	0.3075	18.1748	20.7085	3.475
Max. crit. load (lb)	(Fitness fun. 3)	2160787.2328	27135361.7271	26506061.4521	15580738.5066
No. of increments		6	66	44	32
No. of iteration for last increment					
Mean of fitness values	Fitness fun 1	39228.1598	827420.1266	208118.1808	1741278.4138
	Fitness fun 2	11.5427	56.9995	6.8014	25.9820
	Fitness fun 3	13240749.9747	14083218.3003	23104780.0177	12276148.8053
Standard deviation of fitness values	Fitness fun 1	4961.4366	335190.5445	68211.8514	1074230.8200
	Fitness fun 2	6.0596	22.5369	7.7858	29.0376
	Fitness fun 3	8407213.0250	10888350.1210	22879827.6520	9158817.0309

\*Computed at Apex of Arc Structure



**Figure 14.** Optimal shapes of pyramid structure with 224 bar obtained by use of design variables represented by case I (a), case II (b), case III (c), case IV (d).

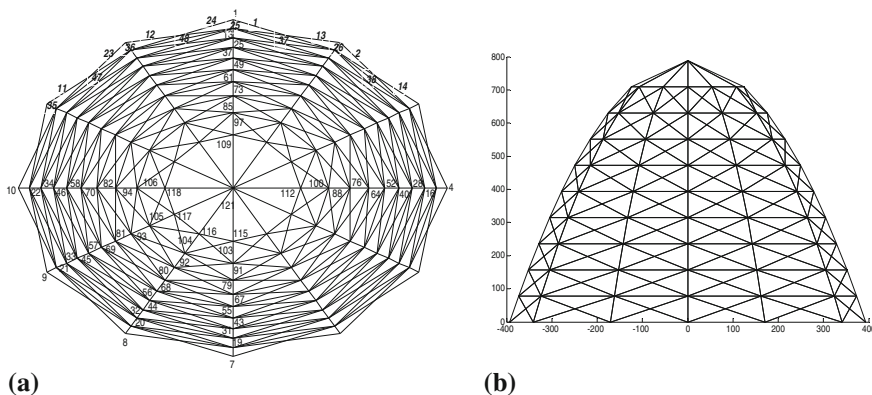


**Figure 15.** Variation of displacement and critical load at apex with weight of pyramid structure with 224 bar.

7.2 Application example 2: Pyramid structure with 224 bar

The pyramid structure has 224 bars. It is originally a single-layer-latticed pyramid (figures 12a and 12b) (Hasancebi & Erbatur 2002). This four-storey pyramid with equal levels is loaded both by two horizontal loads in two x and y directions and a vertical load at its apex. In order to preserve the pyramid form for architectural and aesthetic purposes, nodes located on symmetry axes are restricted to move along these axes. For this purpose, the position of nodes 52, 56, 60 and 64 are fixed to 393.70 in. Equality of certain nodal coordinates is provided by the coordinate parameters, denoted by COOR1-COOR7. The size and shape-related design variables are summarized in table 6.

In order to evaluate the computational performance of PbMIM and PbMOM, their true pareto front and current pareto fronts obtained are presented in figure 13. Also, a statistical output obtained by computing the values of quality-measuring metrics is tabulated in table 7. Considering table 7, it can be said that there is no considerable difference among spread and Igd values of PbMIM and PbMOM for satisfying the required condition as ( $p > 0.050$ ). Furthermore, the Spread and Igd values of PbMIM are lower than PbMOM. Therefore, the quality degree of optimal designations corresponding to PbMIM is higher than PbMOM.



**Figure 16.** Mesh (a) and geometry, (b) attributes of dome structure with 444 bar.

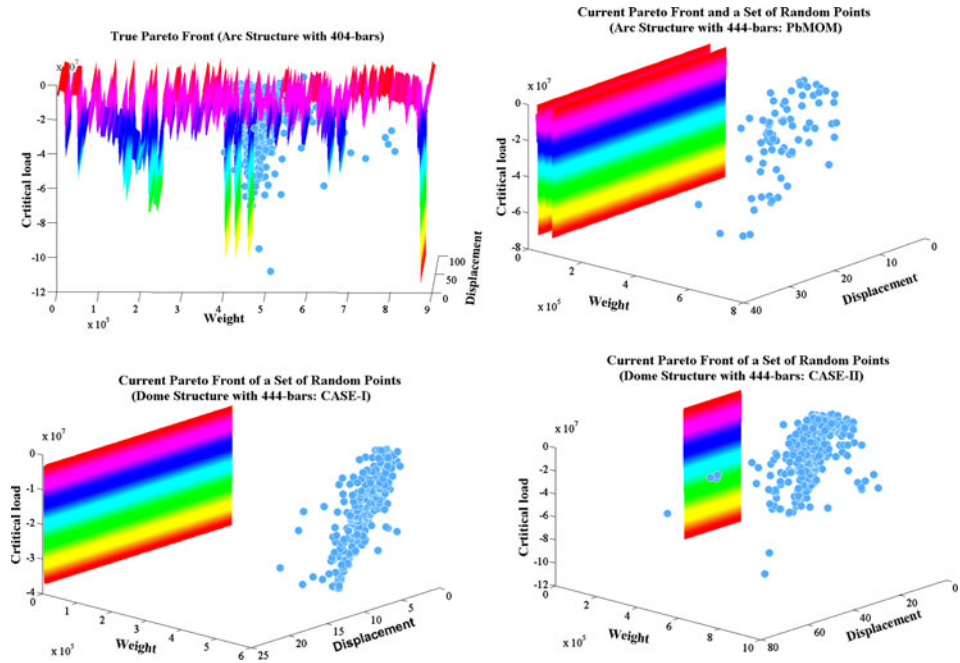
**Table 9.** Optimal designations obtained by execution of PbMIM for dome structure with 444 bar using two design variable sets.

Size (in <sup>2</sup> ) and shape variables (in)	Case I				Case II			
	AI-A12	COOR1	385.3502	AI-A48	121.8856	COOR1	386.6908	COOR1
A13-A24	102.2929	COOR2	31.0044	A49-A96	65.5167	COOR2	36.8082	COOR2
A25-A36	77.1898	COOR3	364.4273	A97-A144	62.5580	COOR3	365.9121	COOR3
A37-A48	34.7586	X22 = Y19	373.6781	A145-A192	77.0829	X22 = Y19	375.4333	X22 = Y19
A49-A60	62.6080	COOR4	93.6668	A193-A240	71.5127	COOR4	66.2061	COOR4
A61-A72	24.1662	COOR5	333.0063	A241-A288	66.5171	COOR5	333.6713	COOR5
A73-A84	59.6318	X34 = Y31	346.8226	A289-A336	65.2560	X34 = Y31	350.6629	X34 = Y31
A85-A96	101.3727	COOR6	167.8464	A337-A384	45.5166	COOR6	154.7827	COOR6
A97-A108	45.6941	COOR7	311.6736	A385-A432	86.1878	COOR7	312.0525	COOR7
A109-A120	60.2131	X46 = Y43	322.5849	A433-A444	87.8159	X46 = Y43	322.8456	X46 = Y43
A121-A132	97.1615	COOR8	227.0188			COOR8	262.6982	COOR8
A133-A144	90.4078	COOR9	275.7064			COOR9	274.7001	COOR9
A145-A156	122.6600	X58 = Y55	296.3544			X58 = Y55	291.7104	X58 = Y55
A157-A168	99.9486	COOR10	322.7595			COOR10	325.8069	COOR10
A169-A180	48.2169	COOR11	257.5766			COOR11	252.2996	COOR11
A181-A192	35.5300	X70 = Y67	265.9443			X70 = Y67	265.1403	X70 = Y67
A193-A204	28.4238	COOR12	365.3104			COOR12	387.0432	COOR12
A205-A216	49.0923	COOR13	227.3600			COOR13	226.7464	COOR13
A217-A228	36.9271	X82 = Y79	241.3365			X82 = Y79	241.1086	X82 = Y79
A229-A240	79.5139	COOR14	406.5609			COOR14	454.6270	COOR14



Table 9. (Continued).

	Case I	Case II
Size (in <sup>2</sup> )	86.9579	188.6380
and shape variables (in)	86.3250	206.5508
A241-A252	76.9888	576.1678
A253-A264	47.2819	156.9422
A265-A276	70.9541	164.7224
A277-A288	24.1934	X106 = Y103
A289-A300	19.6240	COOR15
A301-A312	48.3138	X94 = Y91
A313-A324	46.2379	COOR16
A325-A336	22.1335	COOR17
A337-A348	57.2095	X106 = Y103
A349-A360	70.3271	COOR18
A361-A372	22.2584	COOR19
A373-A384	88.2180	X118 = Y119
A385-A396	82.4277	COOR20
A397-A408	36.0576	Z121
A409-A420	57.7136	
A421-A432		
A433-A444		
	Y10 = Y22 = Y34 = Y46 = Y58 = Y70 = Y82 = Y94 = Y106 = Y118 = Y121 = 0.00	
	X7 = X19 = X31 = X43 = X55 = X67 = X79 = X91 = X103 = X115 = X121 = 0.00	
	COOR1 = Y9*2 = X9 = Y8 = X8*2	COOR15 = Y93*2 = X93 = Y92 = X92*2
	COOR2 = Z7 = Z8 = Z9 = Z10	COOR16 = Z91 = Z92 = Z93 = Z94
	COOR3 = Y21*2 = X21 = Y20 = X20*2	COOR17 = Y105*2 = X105 = Y104 = X104*2
	COOR4 = Z19 = Z20 = Z21 = Z22	COOR18 = Z103 = Z104 = Z105 = Z106
	COOR5 = Y33*2 = X33 = Y32 = X32*2	COOR19 = Y117*2 = X117 = Y116 = X116*2
	COOR6 = Z31 = Z32 = Z33 = Z34	COOR20 = Z115 = Z116 = Z117 = Z118
	COOR7 = Y45*2 = X45 = Y44 = X44*2	
	X10 = Y7 = Y7 = 393.70	
	COOR8 = Z43 = Z44 = Z45 = Z46	
	COOR9 = Y57*2 = X57 = Y56 = X56*2	
	COOR10 = Z55 = Z56 = Z57 = Z58	
	COOR11 = Y69*2 = X69 = Y68 = X68*2	
	COOR12 = Z67 = Z68 = Z69 = Z70	
	COOR13 = Y81*2 = X81 = Y80 = X80*2	
	COOR14 = Z79 = Z80 = Z81 = Z82	



**Figure 17.** True Pareto fronts and random point sets for design example of dome structure with 444 bars.

**Table 10.** Statistical test results computed by use of quality measuring metric values (arc structure with 444 bar).

Multi-objective model	Case no.	Mean (spread)	Std (spread)	P (Welch test)	Mean (Igd)	Std (Igd)	P (Welch test)
PbMIM	1	0.901	0.076	0.746	0.155	0.071	0.657
	2	0.880	0.070		0.153	0.115	
PbMOM	1	0.912	0.058		0.220	0.113	

Mean: average of spread and Igd. Std: standard deviation of spread and Igd

**Table 11.** Genetic output obtained by execution of PbMIM for dome structure with 444 bar (see table 9).

		Case I	Case II
Min. weight (lb)	(Fitness fun. 1)	422251.0074	476924.9852
Min. disp. (in)	(Fitness fun. 2)*	1.5354	0.2522
Max. crit. load (lb)	(Fitness fun. 3)	5723679.9033	1300238.2772
No. of increments		11	5
No. of iteration for last increment		8	10
Mean of fitness values (lb)	Fitness fun 1	487087.6646	475162.3599
	Fitness fun 2	33.0835	11.3302
	Fitness fun 3	42267629.9753	22941818.8341
Standard deviation of fitness values	Fitness fun 1	72168.9213	43863.7239
	Fitness fun 2	10.7712	7.2594
	Fitness fun 3	26609538.1877	16635233.4047

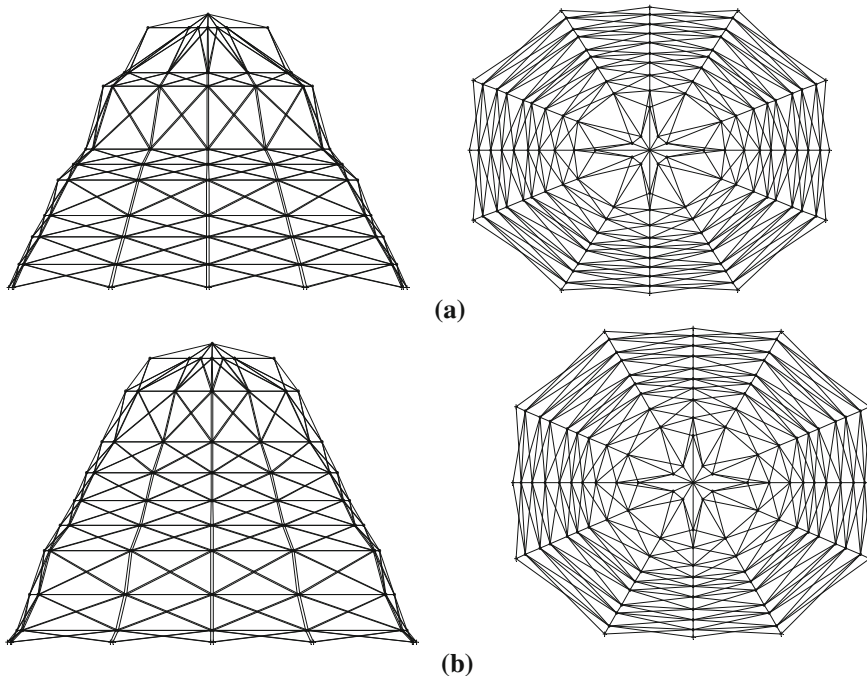
\*Computed at apex of arc structure

Particularly, the success order of four cases obtained by use of PbMIM is Case I, III, II and IV. In this regard, some designations obtained by PbMIM are picked from these cases to carry out a further observation about them. Compared to the other cases, Case I succeed in obtaining better optimal designation with a lower displacement value of 0.3075 in. satisfying the serviceability requirement prescribed in AISC specification ( $393.7 \times 2/300 = 2.6247$ ). However, critical load value corresponding to Case I, 2160787.2328 is poorest compared to a critical load value set of Case II, III and IV (27135361.7271, 26506061.4521 and 15580738.5066) (table 8). It is shown that the final shape of pyramid structure corresponding to Case I has a higher apex height (figure 14). It is clear that computational performance of Case I is higher than the other ones considering the decreased standard deviation value set of fitness values (4961.4366, 6.0596 and 8407213.0250) (table 8) and the higher convergence degree of weight and critical load (figure 15).

The relation between pyramid member cross-sectional areas is also investigated. After a careful examination of pyramid member cross-sectional areas corresponding to Case I (table 6), it is noticed that the diagonal member cross-sectional areas set (24.2942, 21.8630 and 22.4440 for members '1–16', '98–112' and '193–224') are larger than the other member cross-sectional areas set (17.6759, 7.1223, 19.1728 ...etc. for members '17–32', '33–48', '49–64'...etc.). Pyramid member cross-sectional areas are increased towards pyramid support points (table 6).

### 7.3 Application example 3: Dome structure with 444 bar

A dome structure with 444 bar is considered to evaluate the computational performance of PbMIM and PbMOM with respect to an increased number of truss member and severe loading



**Figure 18.** Optimal shapes of dome structure with 444 bar obtained by use of design variables represented by case I (a), case II (b).

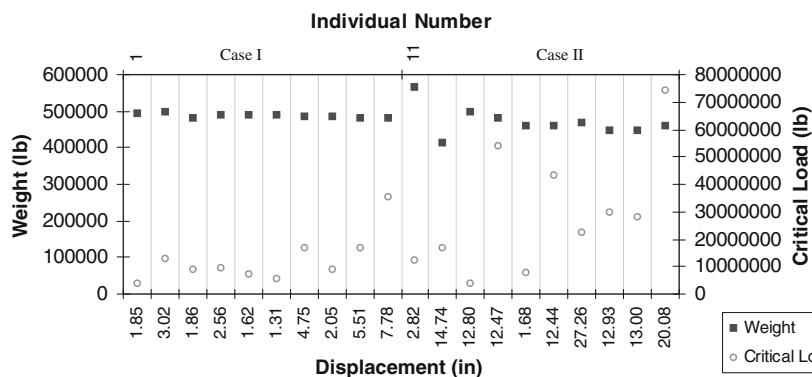
conditions (Lamberti & Pappalettere 2004). It has vertical loads at node 121 and the other free nodes, respectively (see the mesh and geometry attributes of dome structure in figure 16).

While size-related design variables are represented by dome member cross-sectional areas, nodal coordinates are employed for shape-related design variables. Taking the symmetry of dome into account, all nodes collected into groups (COOR1-COOR20) are listed for the quarter part of entire dome structure (table 9). Diagonal and horizontal members at each storey are collected into either separate groups or single group. In this regard, size and shape-related design variables are collected into two combination sets notated by Case I and II (see linked members in table 9).

The true pareto front and current pareto fronts of PbMIM and PbMOM obtained are presented in figure 17. Also, a statistical output obtained by computing the values of quality-measuring metrics is tabulated in table 10. Considering table 10, it can be said that there is no considerable difference among spread and Igd values of PbMIM and PbMOM considering the required condition ( $p > 0.050$ ). Furthermore, the spread and Igd values of PbMIM are lower than PbMOM. Therefore, the quality degree of optimal designations corresponding to PbMIM is higher than PbMOM. Particularly, the success order of two cases obtained by use of PbMIM is Case II and I. In this regard, some designations obtained by PbMIM are picked from these cases to carry out a further observation about them.

The genetic output is listed in table 11. Distinguished from the preceding two examples, Case II achieves to obtain the highest quality of optimal design (476924.9852, 0.2522 and 1300238.2772) satisfying serviceability requirement prescribed in AISC specification ( $400 \cdot 2/300 = 2.6667$ ) for node 121 in z direction. The final shapes obtained for Cases I and II are presented in figure 18. The success of Case II is confirmed by lower standard deviation values (see table 11) and smaller displacement-weight values but higher critical load value (figure 19).

It is clear that there is a relation between diagonal arch member cross-sectional areas and quality degree of optimal designations. In Case I, cross-sectional area values of diagonal arch member (131.2317, 102.2929 for members '1-12' and '13-24') are generally higher than other arch member cross-sectional area set (77.1898 and 34.7586 for members '25-36' and '37-48') (see table 9). Moreover, it is seen that the member cross-sectional areas located in the bottom



**Figure 19.** Variation of displacement and critical load at apex with weight of dome structure with 444 bar.

part of the dome are larger than the member cross-sectional areas located in upper part close to the apex point (table 9).

## 8. Conclusion

In this work, the effect of global stability on design optimization of truss structure is investigated using multiple objective functions. For this purpose, two multi-objective optimization models, the pareto-ranking based multi-objective optimization model (PbMOM) and the pareto-ranking based multi-objective island model (PbMIM) are utilized to optimize the design of the real-world planar and spatial truss structures using different combinations of size and shape-related design variables. In order to compute the nonlinear critical load, arc-length method is employed and improved to estimate the nonlinear critical load with an increased degree of accuracy thereby implementing an eigenvalue analysis into its iteration mechanism.

The following observations are drawn from this work:

- (i) The computational performances of PbMIM and PbMOM are compared by use of two quality-measuring metrics, Spread and Igd. Furthermore, a statistical test is performed to assess the accuracy of these quality-measuring metrics. Although the population size utilized by PbMOM is twice the size of PbMIM's population, it is shown that PbMIM is more efficient optimization tool for optimal design of geometrically-nonlinear truss structures than PbMOM.
- (ii) An increase in the number of members and joints linked causes to decrease the variety in optimal designations. Thus, the quality degree of optimal designations becomes poorer.
- (iii) Diagonal truss members of truss structure have a major role in maximization of critical load. In this regard, it is shown that cross-sectional areas of diagonal truss members corresponding to optimal designations are larger than the other truss members.
- (iv) Considering the optimal designations, it is displayed that member cross-sectional areas located around truss support points are larger than the other part of truss structures.
- (v) It is displayed that PbMIM has a capability of generating feasible designations for even more large and complex design spaces.
- (vi) This study brings a new look at the nonlinearity effect on a simultaneously size and shape optimization of truss structures. Therefore, proposed optimal design procedure deserves more attention. The computational procedures of PbMIM are managed by the probabilistic transition rules. Therefore, size and shape-related design variables are randomly generated. Hence, although it is demonstrated that PbMIM achieves to generate feasible designations, the number of feasible designations is decreased when truss shape obtained is not practically applicable.

The future study will be improved by completing the following lacunae for design optimization of geometrically nonlinear truss structures.

- Shape-related-design variables will be adjusted according to a practically-applicable-truss shape, for example a circle, ellipse, a line with a predefined angle, etc. instead of a random adjustment. Hence, the feasible solutions are correspondingly increased.
- The branched points located on sub-path switched from critical load will be considered to evaluate their effect on the optimality degree.
- The penalization process will be improved to increase the number of feasible designations and the quality of optimal designations.

**Appendix A.** Computation of nodal coordinates.

Node number	Matlab expressions used to compute nodal coordinates of arc structure with 101 bar	
	X Coordinate	Y Coordinate
1	$-(R1*\sin(\pi*(45-\alpha1)/180))$	$(R1*\cos(\pi*(45-\alpha1)/180))$
2	$-((R1+a1)*\sin(\pi*(45-\alpha1)/180))$	$((R1+a1)*\cos(\pi*(45-\alpha1)/180))$
3	$-(R2*\sin(\pi*(45-\alpha2)/180))$	$(R2*\cos(\pi*(45-\alpha2)/180))$
4	$-((R2+a2)*\sin(\pi*(45-\alpha2)/180))$	$((R2+a2)*\cos(\pi*(45-\alpha2)/180))$
5	$-(R3*\sin(\pi*(45-\alpha3)/180))$	$(R3*\cos(\pi*(45-\alpha3)/180))$
6	$-((R3+a3)*\sin(\pi*(45-\alpha3)/180))$	$((R3+a3)*\cos(\pi*(45-\alpha3)/180))$
7	$-(R4*\sin(\pi*(45-\alpha4)/180))$	$(R4*\cos(\pi*(45-\alpha4)/180))$
8	$-((R4+a4)*\sin(\pi*(45-\alpha4)/180))$	$((R4+a4)*\cos(\pi*(45-\alpha4)/180))$
9	$-(R5*\sin(\pi*(45-\alpha5)/180))$	$(R5*\cos(\pi*(45-\alpha5)/180))$
10	$-((R5+a5)*\sin(\pi*(45-\alpha5)/180))$	$((R5+a5)*\cos(\pi*(45-\alpha5)/180))$
11	$-(R6*\sin(\pi*(45-\alpha6)/180))$	$(R6*\cos(\pi*(45-\alpha6)/180))$
12	$-((R6+a6)*\sin(\pi*(45-\alpha6)/180))$	$((R6+a6)*\cos(\pi*(45-\alpha6)/180))$
13	$-(R7*\sin(\pi*(45-\alpha7)/180))$	$(R7*\cos(\pi*(45-\alpha7)/180))$
14	$-((R7+a7)*\sin(\pi*(45-\alpha7)/180))$	$((R7+a7)*\cos(\pi*(45-\alpha7)/180))$
15	$-(R8*\sin(\pi*(45-\alpha8)/180))$	$(R8*\cos(\pi*(45-\alpha8)/180))$
16	$-((R8+a8)*\sin(\pi*(45-\alpha8)/180))$	$((R8+a8)*\cos(\pi*(45-\alpha8)/180))$
17	$-(R9*\sin(\pi*(45-\alpha9)/180))$	$(R9*\cos(\pi*(45-\alpha9)/180))$
18	$-((R9+a9)*\sin(\pi*(45-\alpha9)/180))$	$((R9+a9)*\cos(\pi*(45-\alpha9)/180))$
19	$-(R10*\sin(\pi*(45-\alpha10)/180))$	$(R10*\cos(\pi*(45-\alpha10)/180))$
20	$-((R10+a10)*\sin(\pi*(45-\alpha10)/180))$	$((R10+a10)*\cos(\pi*(45-\alpha10)/180))$
21	0	R11
22	0	(R11+a11)

**Nomenclature**

$p_{ext}$	External joint load
$P$	Load increment used for incremental stage
$p_{int}$	Internal force
$R$	Residual force
$\delta 1$	Displacement increment computed in the end of iteration process (beginning point of incremental stage or first end of arc-length)
$\delta 2$	Displacement increment computed by external load (in incremental stage)
$\delta 3$	Sub-displacement computed in incremental stage but updated in iterative stage
$\delta 4$	Sub-displacement computed using residual force $r$ (in iterative stage)
$\delta 5$	Sub-displacement increment for iterative stage
$\beta$	Scaling factor
$\beta 1$	Sub-scaling factor used to update $\beta$ computed in iterative stage
$\varepsilon$	Desired convergence degree
$inc\_max$	Maximum number of increments
$it\_max$	Maximum number of iterations
$K$	System stiffness matrix

$det[K]$	Determinant of stiffness matrix
$L$	Limit Load
$W$	Weight of truss structure
$\delta$	Deflection
$x, y, z$	Coordinates of nodes
$P$	Material density
$Nda$	Number of different cross-sectional areas
$Nn$	Number of nodes
$Ntm$	Number of truss member

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